NPS55-77-27

NAVAL POSTGRADUATE SCHOOL

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FURTHER CANONICAL METHODS IN THE SOLUTION

OF VARIABLE-COEFFICIENT LANCHESTER-TYPE

EQUATIONS OF MODERN WARFARE

bу

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June 1977

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This report was prepared by:

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This research has been partially supported by the Office of Naval Research.

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SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION	READ INSTRUCTIONS BEFORE COMPLETING FORM							
1. REPORT NUMBER	2. GOVT ACCESSION NO.	3. RECIPIENT'S CATALOG NUMBER						
NPS55-77-27								
4. TITLE (and Subtitie)	5. TYPE OF REPORT & PERIOD COVERED							
Further Canonical Methods in the	Technical Report							
Variable-Coefficient Lanchester-Tof Modern Warfare	6. PERFORMING ORG. REPORT NUMBER							
7. AUTHOR(*)								
	8. CONTRACT OR GRANT NUMBER(*)							
Gerald G. Brown	James G. Taylor Gerald G. Brown							
9. PERFORMING ORGANIZATION NAME AND ADDRESS Naval Postgraduate School		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS						
Monterey, CA 93940								
11. CONTROLLING OFFICE NAME AND ADDRESS	12. REPORT DATE June 1977							
Monterey, CA 93940	Naval Postgraduate School Monterey, CA 93940							
		13. NUMBER OF PAGES 45						
14. MONITORING AGENCY NAME & ADDRESS(If differen	15. SECURITY CLASS, (of this report)							
	Unclassified							
	15a. DECLASSIFICATION/DOWNGRADING							
		JOHEDUCE						
16. DISTRIBUTION STATEMENT (of this Report)								
Approved for public release; distribution unlimited.								
The second of the second secon								
17. DISTRIBUTION STATEMENT (of the abetract entered	in Block 20, if different fro	en Report)						
18. SUPPLEMENTARY NOTES								
19. KEY WORDS (Continue on reverse side if necessary and identify by block number)								
20. ABSTRACT (Continue on reverse side if necessary and identify by block number)								
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LECURITY CLASSIFICATION OF THIS PAGE(When Data Entered)

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FURTHER CANONICAL METHODS IN THE SOLUTION OF VARIABLE-COEFFICIENT LANCHESTER-TYPE EQUATIONS OF MODERN WARFARE

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ABSTRACT

This paper introduces an important new canonical set of functions for solving Lanchester-type equations of modern warfare for combat between two homogeneous forces with power attrition-rate coefficients with "no offset." Tabulations of these functions, which we call Lanchester-Clifford-Schläfli (or LCS) functions, allow one to study this particular variable-coefficient model almost as easily and thoroughly as Lanchester's classic constant-coefficient one. The availability of such tables is pointed out. We show that our choice of LCS functions allows one to obtain important information (in particular, force-annihilation prediction) without having to spend the time and effort to compute force-level trajectories. Furthermore, we show from theoretical considerations that our choice is the best for this purpose. These new theoretical considerations apply in general to Lanchester-type equations of modern warfare and provide guidance for developing other canonical Lanchester functions (i.e. canonical functions for other attrition-rate coefficients). Moreover, our new LCS functions provide valuable information about various related variable-coefficient models. Also, we introduce an important transformation of the battle's time scale that not only many times simplifies the force-level equations but also shows that relative fire effectiveness and intensity of combat are the only two weapon-system parameters determining the course of such variable-coefficient Lanchester-type combat.



O. INTRODUCTION

In an earlier paper [24] we developed some elements of a mathematical theory for solving variable-coefficient Lanchester-type equations of modern warfare for combat between two homogeneous forces and introduced canonical hyperbolic-like Lanchester functions for constructing their solution. Unfortunately, with only these previous results one is limited to computing (however conveniently it may be done) force-level trajectories and cannot gain a real understanding of qualitative model behavior (e.g. force annihilation) without the excessive labor of extensive numerical computations (and only then for specific values of model parameters). Since our earlier work important mathematical discoveries have been made about the qualitative behavior of this combat model, and we wish to show how these new results allow one to parametrically analyze combat modelled by power attrition-rate coefficients (see Section 1 below) with somewhat the same facility as he can study F. W. LANCHESTER's [17] classic constant-coefficient model. In order to obtain this analysis capability, however, one must redefine the Lanchester-Clifford-Schläfli (or LCS) functions, which we introduced in reference 24.

In our earlier paper (see TAYLOR and BROWN [24]) we gave various examples of hyperbolic-like Lanchester functions (in particular, the LCS functions, which arise from power attrition-rate coefficients with "no offset"). Subsequent research by TAYLOR and COMSTOCK [27] has revealed, however, that these canonical LCS functions must be redefined to permit force-annihilation prediction from initial conditions without having to spend the time and effort to compute force-level trajectories. It then became obvious that the entire topic of representing the solution to such Lanchester-type equations in terms of general Lanchester functions (GLF) should be critically reexamined. Thus, the purpose of this paper is to present new general considerations for the selection of canonical Lanchester functions and then apply this theory to the special case of power attrition-rate coefficients with "no offset" (modelling, for example, weapon systems with the same maximum effective range) to obtain new LCS functions. With the availability of tabulations of these new LCS functions

(see Section 6 below), one can study this model almost as easily and thoroughly as

These power Lanchester (i.e. LCS) functions are significant not only because they correspond to attrition-rate coefficients modelling a large class of combat situations of interest but also because they yield valuable information about other related canonical Lanchester functions, e.g. the offset power Lanchester functions (see Note 1 and Section 8 below). This information is, of course, equivalent to knowledge about model behavior (e.g. force annihilation). As a result of our work here one can parametrically analyze variable-coefficient Lanchester-type models for combat between two homogeneous forces with somewhat the same facility as the classic constant-coefficient one. Such models are important for developing insights into the dynamics of combat (see BONDER and HONIG [10], TAYLOR [22], and Section 1 below).

The organization of this paper is as follows. First, we present the variablecoefficient Lanchester-type model that we study in this paper. Next, we discuss the representation of the time history of the force levels for this model in terms of general Lanchester functions (GLF). We show that there are essentially only two kinds of GLF, (I) exponential-like GLF and (II) hyperbolic-like GLF, and that the former (I) provide essential force-annihilation-prediction information about the latter (II). Then we explain why we have chosen to use the hyperbolic-like GLF to construct the model's solution and why the power Lanchester (or LCS) functions introduced by Taylor and Brown $^{[24]}$ must be redefined. Next, we show how the analysis of, for example, the force-level equation is simplified by transforming the independent variable t to normalize the battle's time scale by the intensity of combat. We then introduce our new definition of Lanchester-Clifford-Schläfli functions and show how they arise in solving the transformed X force-level equation. Availability of tabulations of these new LCS functions is discussed, and some uses of the tabulations are illustrated. Finally, insights gained into the dynamics of combat between two homogeneous forces from these developments are discussed.

1. VARIABLE-COEFFICIENT LANCHESTER-TYPE EQUATIONS OF MODERN WARFARE

In this paper we consider the following idealized model for combat between two
homogeneous forces (see Note 2)

$$dx/dt = -a(t)y, dy/dt = -b(t)x, (1)$$

with initial conditions

$$x(t=0) = x_0$$
, and $y(t=0) = y_0$,

where t=0 denotes the time at which the battle begins, x(t) and y(t) denote the numbers of X and Y at time t, and a(t) and b(t) denote time-dependent Lanchester attrition-rate coefficients. We will refer to (1) as variable-coefficient

Lanchester-type equations of modern warfare in honor of the pioneering military operations research work of F. W. Lanchester [17] (see TAYLOR [21] and Taylor and Brown [24]).

Other forms of Lanchester-type equations appear in the literature, but we will not consider these here (see DOLANSKY [14] and Taylor [21]). The Lanchester-type equations (1) yield the X force-level equation

$$d^{2}x/dt^{2} - \{d \ln a(t)/dt\}dx/dt - a(t)b(t)x = 0,$$
(2)

with initial conditions

$$x(t=0) = x_0$$
, and $\{[1/a(t)]dx/dt\}_{t=0} = -y_0$.

Although combat between two military forces is a complex random process, such an idealized deterministic model of the combat attrition process is frequently employed to provide insights into the dynamics of combat (see, for example, BONDER and FARRELL [9], Bonder and Honig [10], TAYLOR and PARRY [28], or WEISS [29]). The reader may consider (1) to model combat in which both sides use aimed fire and target acquisition times are independent of the numbers of firers and targets (see Note 3). New operations research techniques (see, for example, Bonder and Farrell [9], and CLARK [12]) for forecasting temporal variations in fire effectiveness (caused by, for example, changes in force separation, combatant postures, target acquisition rates, etc.) have generated interest in such variable-coefficient combat formulations.

Without loss of generality, we may take a(t) and b(t) to be of the form

$$a(t) = k_{a}g(t),$$
 and $b(t) = k_{b}h(t),$ (3)

where g(t) and h(t) denote the time-varying factors of a(t) and b(t) such that $a(t)/b(t) = k_a/k_b$ for g(t) = h(t). In other words, k_a and k_b denote "scale" factors chosen so that the case of constant coefficients corresponds to g(t) = h(t) =

A large class of tactical situations of interest can be modelled with the following general power attrition-rate coefficients

$$a(t) = k_a (t+K_S)^{\mu}$$
, and $b(t) = k_b (t+K_S+K_0)^{\nu}$, (4)

where K_S , $K_O \ge 0$. The modelling roles of K_S and K_O are discussed in Taylor and Brown [24]. We will call K_S the <u>starting parameter</u>, since it allows us to model (with $\mu, \nu \ge 0$) battles which begin within the maximum effective ranges of the two systems. We will call K_O the <u>offset parameter</u>, since it allows us to model (again, with $\mu, \nu \ge 0$) battles between weapon systems with different effective ranges. Restrictions that must be placed in μ and ν , which are not necessarily integers, are discussed below.

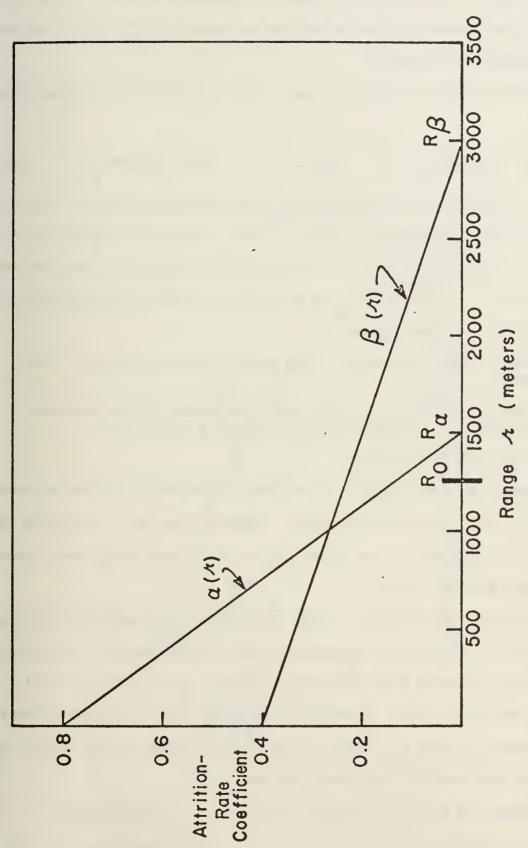
Let us take a few moments to motivate our above notation and further indicate possible applications of our results. Consider BONDER's [4,6] constant-speed attack on a static definsive position modelled by

$$dx/dt = -\alpha(r)y = -\alpha_0(1-r/R_{\alpha})^{\mu}y, \qquad dy/dt = -\beta(r)x = -\beta_0(1-r/R_{\beta})^{\nu}x,$$
 (5)

where $\mu,\nu\geq 0$ and R_{α} denotes the maximum effective range of the Y weapon system. Then the starting parameter and the offset parameter are given by

$$K_S = (R_\alpha - R_0)/v$$
, and $K_O = (R_\beta - R_\alpha)/v$, (6)

where R_0 denotes the battle's opening range and v > 0 denotes the constant attack speed. Hence, $K_0, K_S \ge 0 \Leftrightarrow R_\beta \ge R_\alpha \ge R_0$. By considering (6) and Figure 1, the reader should have no trouble in understanding our terminology for K_S and K_0 . In the model (5) μ , for example, is used to model the range dependence of Y's attrition-rate



coefficients modelling constant-speed attack. [Notes: 1. The maximum effective ranges of denoted as R_0 and (as shown) R_0 < minimum (R_α, R_β) . 3. The offset parameter is given by Figure 1. Explanation of offset parameter K_0 and starting parameter $K_{
m S}$ for power attrition-rate the two weapon systems are denoted as $_{\alpha}^{}$ and $_{\beta}^{}$. 2. The opening range of battle is $K_0 = (R_\beta - R_\alpha)/v$. 4. The starting parameter is given by $K_S = (R_\alpha - R_0)/v$.]

coefficient (see Figure 2). Observing that range is related to time by $r(t) = R_0 - vt$, we readily see that the longest the battle can last is given by $t_{max} = R_0/v$, at which time zero force separation is reached.

When the offset parameter is equal to zero (i.e. $K_0 = 0$), then the coefficients (4) reduce to

$$a(t) = k_a (t+K_g)^{\mu}$$
, and $b(t) = k_b (t+K_g)^{\nu}$. (7)

We will refer to (7) as <u>power attrition-rate coefficients with "no offset."</u> The purpose of this paper is to extend our previous results^[24] and introduce new power Lanchester functions that allow more information to be more conveniently extracted from the model (1) with coefficients (7). Specifically, one would want to obtain information such as:

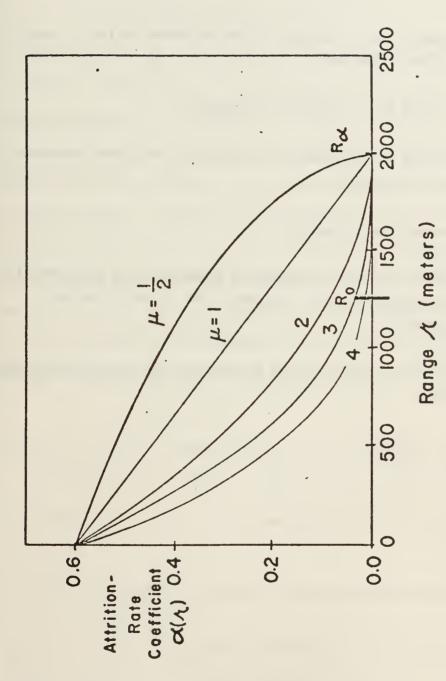
- (01) Who will "win"? Be annihilated?
- (Q2) How do force levels decrease over time and how many survivors will the winner have?
- (Q3) How do changes in the initial force levels and weapon system parameters affect the outcome? Is concentration of forces a good tactic?
- (Q4) How long will the battle last?

To conveniently answer questions (Q1), (Q3), and (Q4) a redefinition of the Lanchester-Clifford-Schläfli (or LCS) functions is required. Moreover, not only are results for the coefficients (7) of interest in their own right but they also provide much valuable information about the general case (4).

2. REPRESENTATION OF SOLUTION IN TERMS OF GENERAL LANCHESTER FUNCTIONS

In this section we discuss how to construct the solution to the X force-level equation (2) in terms of certain basic building blocks that we have chosen to call general Lanchester functions (GLF). We feel that these GLF should be chosen according to the guidelines shown in Table I. Special cases of these general considerations have been given by Taylor and Brown [24] and Taylor and Comstock [27].

Let us introduce the GLF
$$\mathbf{x}^{T} = (\mathbf{x}_{1} \ \mathbf{x}_{2})$$
 and $\mathbf{y}^{T} = (\mathbf{y}_{1} \ \mathbf{y}_{2})$ which satisfy
$$\dot{\mathbf{x}} = ka(t)L\mathbf{y}, \qquad \dot{\mathbf{y}} = (1/k)b(t)L\mathbf{x}, \qquad (8)$$



maximum effective range of the system is denoted as R = 2000 meters. 2. $\alpha(r=0) = \alpha_0 = 0.6 \text{ X}$ zero force separation (range). 3. The opening range of battle is denoted as R_0 = 1250 meters Dependence of the attrition-rate coefficient $\alpha(r)$ on the exponent μ with maximum effective casualties/(unit time x number of Y units) denotes the Y force weapon system kill rate at range of the weapon system and kill capability at zero range held constant. [Notes: 1. and (as shown) $R_0 < R$.] Figure 2.

TABLE I. Requirements for General Lanchester Functions

- (R1): They can be used to construct the solutions to the X and Y force-level equations.
- (R2): They should be as "simple" as possible.
- (R3): A given set of functions should apply to as large a class of battles as possible.
- (R4): They should be nonnegative.
- (R5): They should reduce to elementary transcendental functions in special cases such as a constant ratio of attrition-rate coefficients.
- (R6): They should provide as much information as possible about model behavior.

where without further specification k may be any positive constant and L is any 2×2 square matrix such that $L^2 = I$. [The initial conditions for (8) are to be chosen so that the requirements of Table I are met and are discussed below.] Then $\underset{\sim}{\times}$ satisfies the vector equation

$$\ddot{x} - \{d \ln a(t)/dt\}\dot{x} - a(t)b(t)\dot{x} = 0.$$
 (9)

In other words, both x_1 and x_2 satisfy the X force-level equation (2), and similarly y_1 and y_2 satisfy the Y force-level equation.

Let us now investigate all the possible forms for the above GLF. Intuitively, we would expect two possibilities (keeping the requirements of Table I in mind): exponential-like functions (one strictly increasing and the other strictly decreasing) and hyperbolic-like functions. [We are reminded of these two possibilities by the well-known constant-coefficient results.] Two such types of GLF appear in Taylor and Brown [24] (hyperbolic-like functions) and in Taylor and Comstock [27] (exponential-like functions). We will show that these are the only two possibilities if the requirements of Table I are to met and show the relationship between these two types of GLF.

It is readily shown that any 2×2 matrix such that $L^2 = I$ must take one of five forms.

LEMMA 1: If
$$L^2 = I$$
, then L must take one of the following five forms: (A) $\begin{pmatrix} \alpha & \beta \\ [1-\alpha^2]/\beta & -\alpha \end{pmatrix}$ with $\beta \neq 0$, (B) $\begin{pmatrix} 1 & 0 \\ \gamma & -1 \end{pmatrix}$, (C) $\begin{pmatrix} -1 & 0 \\ \gamma & 1 \end{pmatrix}$, (D) L = I, or

(E) L = -I, where α , β , and γ are unrestricted with the exception that $\beta \neq 0$.

If in addition $L = L^T$ and |L| = -1, then L is an orthogonal matrix and must be of the form

$$L = \begin{pmatrix} \cos \phi & \sin \phi \\ \sin \phi & -\cos \phi \end{pmatrix}. \tag{10}$$

It seems reasonable to give our Lanchester functions symmetry by requiring that $L = L^{T}$. We observe that the hyperbolic-like GLF of Taylor and Brown [24] correspond to

 $L_{\rm H} = \begin{pmatrix} 0 & 1 \\ 1 & 0 \end{pmatrix}$. In order that (R4) of Table I be met for a constant ratio of attrition-rate coefficients, i.e.

$$a(t) = k_{a}h(t)$$
, and $b(t) = k_{b}h(t)$, (11)

we know (see Taylor and Brown [24]) that we should choose $k = \sqrt{k_b/k_a}$ so that

$$\dot{x} = \sqrt{k_b/k_a} \ a(t)Lx, \qquad \dot{y} = \sqrt{k_a/k_b} \ b(t)Lx. \tag{12}$$

The general solution to the X force-level equation (2) is given by

$$x(t) = c^{T}x. (13)$$

Introducing the 2×2 matrix

$$S(t) = \begin{pmatrix} x^{T}(t) \\ y^{T}(t)L^{T} \end{pmatrix}, \qquad (14)$$

we obtain from the initial conditions to (2) that

$$C = S^{-1}(t=0) \begin{pmatrix} x_0 \\ -\sqrt{k_a/k_b} & y_0 \end{pmatrix}$$
 (15)

Observing that

$$\frac{d}{dt} |S(t)| = \begin{vmatrix} \dot{x}^{T}(t) \\ \dot{x}^{T}(t)L^{T} \end{vmatrix} + \begin{vmatrix} \dot{x}^{T}(t) \\ \dot{y}^{T}(t)L^{T} \end{vmatrix},$$

we readily see that $|S(t)| = \text{constant } \forall t \epsilon(t_0, +\infty)$, where |S| denotes the determinant of the square matrix S. Thus, we may take

$$|S(t)| = constant.$$
 (16)

Let us also observe that

$$\sqrt{k_b/k_a} \ a(t) |S(t)| = W(x_1, x_2),$$
 (17)

where $W(x_1, x_2)$ denotes the Wronskian of x_1 and x_2 .

Now let us subject the fundamental system of solutions $\underset{\sim}{\textbf{x}}$ to linear transformation

$$\hat{\mathbf{x}} = \mathbf{A}\mathbf{x},\tag{18}$$

such that the form of the equations (12) remains invariant, i.e.

$$\hat{\hat{\mathbf{x}}} = \sqrt{k_b/k_a} \, a(t) \, \hat{\mathbf{L}} \, \hat{\mathbf{y}}, \qquad \qquad \hat{\hat{\mathbf{y}}} = \sqrt{k_a/k_b} \, b(t) \, \hat{\mathbf{L}} \, \hat{\mathbf{x}}, \qquad (19)$$

where \hat{L} again is such that \hat{L}^2 = I. If \hat{L} is given, it follows that

$$\hat{\mathbf{y}} = \hat{\mathbf{L}} \mathbf{A} \mathbf{L} \mathbf{y}. \tag{20}$$

Furthermore

$$\hat{S}^{T}(t) = AS^{T}(t), \qquad (21)$$

so that

$$|\hat{S}(t)| = |A| \cdot |S(t)| = constant.$$
 (22)

We also observe that $W(\hat{x}_1, \hat{x}_2) = |A|W(x_1, x_2)$. Considering the quotient of the two general Lanchester X-functions (GLXF)

$$\eta(t) = x_1/x_2,$$
 (23)

ve see that under the linear transformation (18) we have

$$d\hat{\eta}/dt = \{|A|/(a_{12}\eta + a_{22})^2\}d\eta/dt.$$
 (24)

We now show that the only possible GLF that satisfy the requirements of Table I corresponding to $L = L^T$) are the exponential ones shown in Table II and the hyperolic ones shown in Table III. According to Lemma 1 if $L^2 = I$, then L must take one of five forms. It is impossible to have L = -I and satisfy (R4) of Table I (see lote 4). If L = I and we try to specify the "simplest" initial conditions [i.e. specify initial conditions such that the GLF take the "simplest" form (satisfy (R2) of Table I)], we find that we may take L to have one of the three remaining forms (see lote 5). If we require that L be symmetric for simplicity [requirement (R2)], then is an orthogonal matrix with the form (10). If in (10) we take $\cos \phi$ and $\sin \phi$ equal to -1, 0, or 1 in order that the GLF take the "simplest" form, we find that the only two distinct possibilities for L are L_E and L_H as given in Tables II and III (see argument given in Note 5). Thus, we have shown that if we wish to construct the solution (13) to (2) by using GLF with the properties given in Table I,

here are essentially only two possibilities: the exponential-like GLF introduced by

Table II. Exponential-Like General Lanchester Functions

1.
$$x_{E}^{T}(t) = (E_{X}^{+}(t;Q*) \quad E_{X}^{-}(t;Q*)), \quad x_{E}^{T}(t) = (E_{Y}^{+}(t;Q*) \quad E_{Y}^{-}(t;Q*))$$

2.
$$x_E^T(t=t_0) = (1/0*1), x_E^T(t=t_0) = (1 0*)$$

$$L_{E} = \begin{pmatrix} 1 & 0 \\ 0 & -1 \end{pmatrix}$$

4.
$$|S_{E}(t)| = -2$$

Table III. Hyperbolic-Like General Lanchester Functions

1.
$$\chi_{H}^{T}(t) = (S_{\chi}(t) C_{\chi}(t)), \quad \chi_{H}^{T}(t) = (S_{\chi}(t) C_{\chi}(t))$$

2.
$$x_H^T(t=t_0) = (0 \ 1), \quad x_H^T(t=t_0) = (0 \ 1)$$

$$L_{H} = \begin{pmatrix} 0 & 1 \\ 1 & 0 \end{pmatrix}$$

4.
$$|S_{H}(t)| = -1$$

Taylor and Comstock [27] and the hyperbolic-like GLF introduced by Taylor and Brown [24].

We call the quantity Q*, which appears in the initial conditions for the exponential-like GLF, the parity condition parameter. It is chosen so that (see Note 6)

$$E_{X}^{-}(t;Q^{*}), \quad E_{Y}^{-}(t;Q^{*}) > 0 \quad \text{for all finite } t \ge t_{0}.$$
 (25)

It may be considered to be the enemy force equivalent of a friendly X force of unit strength. Taylor and Comstock^[27] show how knowledge of the parity condition parameter allows one to predict force annihilation from initial conditions without explicitly computing force-level trajectories. We observe that the exponential-like GLF cannot be computed until one has solved the associated auxiliary parity-condition problem^[27] (i.e. knows how to predict force-annihilation). For this reason and others (see Taylor and Comstock^[27]), the exponential-like GLF are mainly of theoretical importance. Moreover, in the next paragraph we show how the exponential-like GLF provide valuable force-annihilation information about the hyperbolic-like GLF.

We now show that the limiting value of the quotient of the two hyperbolic-like GLXF, $n_{\rm H} = S_{\rm X}/C_{\rm X}$, is equal to the reciprocal of the parity condition parameter, i.e. (30) holds. We know that the two types of GLF are related by a linear transformation

$$x_{H} = Ax_{E}. (26)$$

From (21) at t = 0, we have

$$\begin{pmatrix} 0 & 1 \\ 1 & 0 \end{pmatrix} = A \begin{pmatrix} 1/Q * & 1 \\ 1 & -Q * \end{pmatrix},$$

whence

$$A = \begin{pmatrix} 1/2 & -1/(2Q^*) \\ 0^*/2 & 1/2 \end{pmatrix}. \tag{27}$$

Considering (23) and (26), we see that

$$\eta_{H}(t) = \{a_{11}\eta_{E}(t;Q^*) + a_{12}\}/\{a_{21}\eta_{E}(t;Q^*) + a_{22}\}.$$
 (28)

Recalling that $\eta_{E}(t;Q^{*}) = E_{X}^{+}(t;Q^{*})/E_{X}^{-}(t;Q^{*})$ and $\lim_{t\to +\infty} E_{X}^{-}(t;Q^{*}) = 0$, we see that

$$\lim_{t \to +\infty} \eta_{E}(t; Q^{*}) = +\infty, \tag{29}$$

so that by (27) and (28)

$$\lim_{t \to +\infty} \eta_{H}(t) = 1/Q^*. \tag{30}$$

This result (30) is highly significant because it leads to a computation method for determining Q*. Moreover, in the future we will show how the LCS functions introduced in this paper (see Section 5 below) play a crucial role in such numerical determinations. Let us further note that (24) reads

$$d\eta_{H}/dt = {2/(Q*\eta_{E}+1)^{2}}d\eta_{E}/dt,$$
 (31)

so that both $n_E(t)$ and $n_H(t)$ are strictly increasing functions of t, since $dn_E/dt = 2\sqrt{k_b/k_a} a(t)/\{E_X^-(t;Q^*)\}^2$.

3. CONSIDERATIONS FOR THE CHOICE OF GENERAL LANCHESTER FUNCTIONS

In Table I we give the general requirements that we feel should be placed upon

GLF. These requirements are motivated by the properties possessed by the functions

(namely, the exponential and hyperbolic ones) that one uses to construct the solution

to (1) in the constant-coefficient case.

We specify (R3) so that as few tabulations of GLF as possible will be required. Consequently, we specify the initial conditions for the GLF at $t_0 = \max(t_0^X, t_0^Y)$, where t_0^X denotes the largest finite singular point on the t-axis for the X force-level equation (2) (see reference 24). Thus, a(t) and b(t) are positive continuous functions $\forall t \in (t_0, +\infty)$. Since at most one of x(t) and y(t) can vanish in $[0, +\infty)$ (see Note 7), we have what the mathematician calls a nonoscilliatory solution to (1). In this case we can construct the solution to (2) out of nonnegative components and will find it convenient to do so.

As we have shown above, there are essentially only two types of GLF that satisfy the requirements of Table I: the exponential-like GLF of Table II and the hyperbolic-like GLF of Table III. We feel, however, that the hyperbolic-like functions are to be preferred for two reasons: (1) they apparently are more convenient for

parametric studies in which one might, for example, want to vary initial force levels or some measure of relative fire effectiveness ^[24]; and (2) accurate values of the exponential-like GLF are, in general, difficult (in fact, essentially impossible for large values of t) to determine, since their initial conditions depend on the parity condition parameter Q* (see Note 8). In terms of the hyperbolic-like GLF, the solution to (2) is given by ^[24]

 $x(t) = x_0 \{C_Y(t=0)C_X(t) - S_Y(t=0)S_X(t)\} - y_0 \sqrt{k_a/k_b} \{C_X(t=0)S_X(t) - S_X(t=0)C_X(t)\}. \tag{32}$ We observe that for $t_0 < 0$, for example, $C_X(t=0) > 1$ and $S_X(t=0) > 0$ so that except for the quasi-autonomous case in which a(t)/b(t) = constant (see Note 9), the solution (32) only simplifies when $t_0 = 0$ (see Theorem 1 of Taylor and Brown [24]) (see Note 10).

Unfortunately, the power Lanchester (or LCS) functions introduced by Taylor and Brown [24] were inappropriately defined to yield all the information sought about the combat model (1) with power attrition-rate coefficients (7). In particular, the time at which a side will be annihilated cannot be determined (without the explicit calculation of the entire force-level trajectories) from the initial conditions. Subsequent work by Taylor and Comstock has yielded a theory of force-annihilation prediction [2] The purpose of the paper at hand is to redefine the power Lanchester functions in light of these subsequent results. We also thought it important to present the general considerations behind this selection of canonical Lanchester functions.

Moreover, the form of the LCS functions is simplified and insight gained into the dynamics of combat by transforming the battle's time scale. Thus, certain transformations of variables may be desirable in the development of hyperbolic-like GLF, and the specifications of Table III should be interpreted as being "symbolic" and not taken literally.

4. A TRANSFORMATION TO NORMALIZE THE BATTLE'S TIME SCALE BY THE INTENSITY OF COMBAT

Let $\int_{-\infty}^{t}$ denote an indefinite integral, denote the relative effectiveness as R(t), i.e.

$$R(t) = b(t)/a(t), \tag{33}$$

and let K be an arbitrary constant to be conveniently chosen. Then Theorem 2 of Taylor and Brown [24] may be stated as follows.

THEOREM 1: A necessary and sufficient condition to be able to transform the X force-level equation (2) by a transformation of the independent variable t into a linear second order ordinary differential equation with constant coefficients is that

$$[d\{l_n R(t)\}/dt]/\sqrt{a(t)b(t)} = CONSTANT,$$

and then the desired substitution is given by

$$\tau = K \int_{a(s)b(s)}^{t} ds.$$

We observe that Theorem 1 says that we can transform the X force-level equation to a constant coefficient one if and only if $d\{\ln R(t)\}/d\tau = \text{CONSTANT}$. We also assume that the following condition holds.

If Condition (A) is to hold, then for the power attrition-rate coefficients (7) we must have $\mu, \nu > -1$.

Motivated by both Theorem 1 and the well-known constant-coefficient results, we introduce the new independent variable τ defined by

$$\tau = \int_{t_0}^{t} \sqrt{a(s)b(s)} ds.$$
 (34)

By Condition (A) and the Cauchy-Schwarz inequality for integrals (see p. 123 of BELLMAN $^{[3]}$ the integral in (34) is well defined (i.e. bounded). The transformation (34) has an inverse $t(\tau)$, since $d\tau/dt > 0 \ \forall \ t > t_0$. We also define

$$\tau_0 = \tau(t=0). \tag{35}$$

We observe that for $t_0 \le 0$ we have $\tau_0 \ge 0$. Recalling the constant-coefficient results we will call the quantity $\sqrt{a(t)b(t)}$ the "intensity of combat" (see also Taylor and Parry [28]); since the larger it is, the more quickly the battle is moving towards termination. The average intensity of combat is given by $\sqrt{a(t)b(t)} = (1/t) \int_0^t \sqrt{a(s)b(s)} \, ds$. Then we have

$$\tau - \tau_0 = \{ (1/t) \int_0^t \sqrt{a(s)b(s)} \, ds \}_t = \sqrt{a(t)b(t)} \, t.$$
 (36)

The substitution (34) transforms (2) into

$$d^{2}x/d\tau^{2} + (1/2) \{d \ln R(t)/d\tau\} dx/d\tau - x = 0,$$
 (37)

with initial conditions

$$x(\tau=\tau_0) = x_0$$
, and $\{R^{1/2}(t)dx/d\tau\}_{\tau=\tau_0} = -y_0$.

Theorem 1 tells us that unless (37) is a constant-coefficient equation, it is impossible to transform the X force-level equation (2) into a constant-coefficient equation by a transformation of the independent variable alone. Also, equation (37) is highly significant because it clearly shows us that the course of combat depends on just two weapon-system parameters: (1) R(t) = b(t)/a(t), the relative fire effectiveness (X to Y) of the two combatants, and (2) $I(t) = \sqrt{a(t)b(t)}$, the intensity of combat (through equation (34), which relates I(t) to τ). Both these parameters may vary over time. In particular, from (37) we see that the nature of temporal variations in relative fire effectiveness will have a significant effect upon the course of combat.

For the power attrition-rate coefficients with no offset (7), the transformed X force-level equation (37) becomes

$$d^{2}x/d\tau^{2} + \{(2q-1)/\tau\}dx/d\tau - x = 0,$$
 (38)

with initial conditions

$$\mathbf{x}(\tau=\tau_0) = \mathbf{x}_0$$
, and $\{(\tau/2)^{2q-1}d\mathbf{x}/d\tau\}_{\tau=\tau_0} = -\mathbf{y}_0\sqrt{k_a/k_b}(\sqrt{k_ak_b}/(\mu+\nu+2))^{2q-1}$,

where $q = (v+1)/(\mu+v+2)$ and

$$\tau = \tau(t) = (2\sqrt{k_a k_b}/(\mu + \nu + 2))(t + K_S)^{(\mu + \nu + 2)/2}.$$
 (39)

Hence, $\tau_0 = (2\sqrt{k_a k_b}/(\mu+\nu+2))K_S^{(\mu+\nu+2)/2}$. Let us observe that $\forall \mu, \nu > -1$ we have 0 < q < 1. Furthermore, $q > 1/2 \Leftrightarrow dR/dt > 0$, i.e. R(t) is a strictly increasing function of time.

5. LANCHESTER-CLIFFORD-SCHLÄFLI FUNCTIONS

Consider the function $F_{\alpha}(x)$ defined by the power series

$$F_{\alpha}(x) = \Gamma(\alpha) \sum_{k=0}^{\infty} (x/2)^{2k} / \{k! \Gamma(k+\alpha)\}. \tag{40}$$

For $\alpha \neq 0,-1,-2,\ldots$ the radius of convergence for $F_{\alpha}(x)$ is infinite by the ratio test for convergence of power series (see, for example, KNOPP^[16]). Hence, $F_{\alpha}(z)$ is an entire function of the complex variable z = x + iy with an essential singularity at the point at infinity. Now consider the function $H_{\alpha}(x)$ defined by the infinite series

$$H_{\alpha}(x) = \Gamma(\alpha) \sum_{k=0}^{\infty} (x/2)^{2(k+\alpha)} / \{k! \Gamma(k+\alpha+1)\}.$$
 (41)

Observing that

$$H_{\alpha}(x) = (1/\alpha)(x/2)^{2\alpha} F_{\alpha+1}(x),$$
 (42)

we see that for $\alpha > 0$ the infinite series (41) is uniformly convergent on compact subsets of the complex plane. From (42) we can readily deduce the recursive relation

$$F_{\alpha}(x) = F_{\alpha+1}(x) + \{(x/2)^2 / [\alpha(\alpha+1)]\} F_{\alpha+2}(x). \tag{43}$$

We will call the functions $F_{\alpha}(x)$ and $H_{\alpha}(x)$ <u>Lanchester-Clifford-Schläfli</u> functions (see Note 11). Other properties are readily deduced and are given in Table IV.

Table IV. Properties of the LCS Functions $F_{\alpha}(x)$ and $H_{\alpha}(x)$.

1.
$$dF_{\alpha}/dx = (x/2)^{1-2\alpha}H_{\alpha}(x)$$

2.
$$dH_{\alpha}/dx = (x/2)^{2\alpha-1}F_{\alpha}(x)$$

3.
$$F_{\alpha}(x)F_{1-\alpha}(x) - H_{\alpha}(x)H_{1-\alpha}(x) = 1 \quad \forall x$$
 where α is not an integer (including zero)

4.
$$F_{\alpha}(x=0) = 1$$

5.
$$H_{\alpha}(x=0) = 0$$
 for $\alpha > 0$

$$6. dF_{\alpha}/dx(x=0) = 0$$

7.
$$\{(x/2)^{1-2\alpha} dH_{\alpha}/dx\}_{x=0} = 1$$

8.
$$F_{1/2}(x) = \cosh x$$

9.
$$H_{1/2}(x) = \sinh x$$

The function $F_{\alpha}(x)$ satisfies the second order ordinary differential equation

$$d^{2}F_{\alpha}/dx^{2} + \{(2\alpha - 1)/x\}dF_{\alpha}/dx - F_{\alpha} = 0,$$
 (44)

with initial conditions

$$F_{\alpha}(x=0) = 1$$
, and $dF_{\alpha}/dx(x=0) = 0$,

while $H_{\alpha}(x)$ satisfies

$$d^{2}H_{\alpha}/dx^{2} - \{(2\alpha - 1)/x\}dH_{\alpha}/dx - H_{\alpha} = 0,$$
 (45)

with initial conditions (for $\alpha > 0$)

$$H_{\alpha}(x=0) = 0$$
, and $\{(x/2)^{1-2\alpha}dH_{\alpha}/dx\}_{x=0} = 1$.

Thus, $\{F_{\alpha}, H_{1-\alpha}\}$ is a fundamental system of solutions to

$$d^{2}F/dx^{2} + \{(2\alpha - 1)/x\}dF/dx - F = 0, \tag{46}$$

with Wronskian $W(F_{\alpha}, H_{1-\alpha}) = (x/2)^{1-2\alpha}$. Let us observe that (see Table III)

$$C_{X}(t) = F_{q}(\tau(t)), \quad S_{X}(t) = {\sqrt{k_{a}k_{b}}/(\mu+\nu+2)}^{2q-1}H_{p}(\tau(t)),$$
 (47)

$$C_{Y}(t) = F_{p}(\tau(t)), \quad S_{Y}(t) = {\sqrt{k_{a}k_{b}}/(\mu+\nu+2)}^{1-2q}H_{q}(\tau(t)),$$
 (48)

where p = 1 - q. If we define

$$T_{\alpha}(x) = H_{1-\alpha}(x)/F_{\alpha}(x),$$
 (49)

then

$$\eta_{H}(t) = T_{X}(t) = S_{X}(t)/C_{X}(t) = {\sqrt{k_{a}k_{b}}/(\mu+\nu+2)}^{2q-1}H_{p}(\tau(t))/F_{q}(\tau(t)), \quad (50)$$

where $T_X(t)$ denotes a hyperbolic-like GLF. Observing that $\lim_{t\to +\infty} \tau(t) = +\infty$, we see $t\to +\infty$

that $T_{\alpha}(x)$ is a strictly increasing function of $x \ \forall \ x \in [0,+\infty)$ and

$$0 \le T_{\alpha}(x) < \Gamma(1-\alpha)/\Gamma(\alpha) \qquad \text{for} \quad 0 \le x < +\infty, \tag{51}$$

with

$$\lim_{x \to +\infty} T_{\alpha}(x) = \Gamma(1-\alpha)/\Gamma(\alpha), \qquad (52)$$

since by the value of Q^* determined by Taylor and Comstock^[27] for the power attrition-rate coefficients (7), denoted as $Q^*(\mu,\nu,K_0=0)$, we have (see (30) and (50))

$$\lim_{t\to +\infty} T_{X}(t) = 1/Q*(\mu,\nu,K_{0}=0) = \{\sqrt{k_{a}k_{b}}/(\mu+\nu+2)\}^{2q-1}\Gamma(p)/\Gamma(q).$$

Comparing (38) and (46), we see that the solution to (38) is given by

$$x(t) = x_0 \{ F_p(\tau_0) F_q(\tau(t)) - H_q(\tau_0) H_p(\tau(t)) \}$$

$$-y_0 \sqrt{k_a/k_b} \{ \sqrt{k_a k_b} / (\mu + \nu + 2) \}^{2q-1} \{ F_q(\tau_0) H_p(\tau(t)) - H_p(\tau_0) F_q(\tau(t)) \}.$$
(53)

The time to annihilate X is determined by $x(t=t_y^a) = 0$ and thus

$$T_{q}(\tau(t_{X}^{a})) = \{x_{0}F_{p}(\tau_{0}) + y_{0}\sqrt{k_{a}/k_{b}}(\sqrt{k_{a}k_{b}}/(\mu+\nu+2))^{q-p}H_{p}(\tau_{0})\}/$$

$$\{x_{0}H_{q}(\tau_{0}) + y_{0}\sqrt{k_{a}/k_{b}}(\sqrt{k_{a}k_{b}}/(\mu+\nu+2))^{q-p}F_{q}(\tau_{0})\},$$
(54)

where $T_{\alpha}(\tau)$ is given by (49) and from (51)

$$0 \le T_{q}(\tau) < \Gamma(p)/\Gamma(q). \tag{55}$$

For $t_0 = -K_S = 0$, (54) simplifies to

$$T_{q}(\tau(t_{X}^{a})) = (x_{0}/y_{0})\sqrt{k_{b}/k_{a}}(\sqrt{k_{a}k_{b}}/(\mu+\nu+2))^{p-q}.$$
 (56)

From (54) and (55) we may deduce the following theorem:

THEOREM 2: Consider combat between two homogeneous forces described by (1) with power attrition-rate coefficients (7). Assume that these equations hold for all time and that Y "wins" when $x(t_f) = 0$ with $y(t_f) > 0$. Then Y will win if and only if

$$\Gamma(q) \{x_0 F_p(\tau_0) + y_0 \sqrt{k_a/k_b} (\sqrt{k_a k_b}/(\mu+\nu+2))^{q-p} H_p(\tau_0)\} < 0$$

$$\Gamma(p) \{x_0^{}H_q^{}(\tau_0^{}) + y_0^{}\sqrt{k_a^{}/k_b^{}}(\sqrt{k_a^{}k_b^{}}/(\mu + \nu + 2))^{q-p}F_q^{}(\tau_0^{})\}.$$

For $t_0 = 0$ (i.e. $K_S = 0$ and $\tau_0 = 0$), Y will win if and only if $x_0 \Gamma(q) < y_0 \sqrt{k_a/k_b} (\sqrt{k_a k_b}/(\mu + \nu + 2))^{q-p} \Gamma(p).$

6. TABULATIONS OF LCS FUNCTIONS

Tabulations of the Lanchester-Clifford-Schläfli functions are available in two of the authors' reports, also available from the National Technical Information Service (see references 25 and 26). These reports contain five-decimal-place tables of the hyperbolic-like LCs functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for values of the argument x=0.00(0.01)2.00(0.1)10.0 and various values of the order α . The short table [25] contains tabulations for $\alpha=1/2,1/3,2/3,1/4,3/4,1/5,2/5,3/5,4/5,3/7$, and 4/7 corresponding to $\mu,\nu=0,1,2,3$; while the longer table [26] contains tabulations for $\alpha=1/2,1/3,2/3,1/4,3/4,1/5,2/5,3/5,4/5,2/7,3/7,4/7,5/7,4/9,5/9,3/11,5/11,6/11,8/11,5/13,8/13,5/17,12/17,5/21, and 16/21 corresponding to <math>\mu,\nu=0,1/4,1/2,1,1$ 1/2,2,3. As we have seen above in Section 1 (see (5) and Figure 2), such values of μ and ν allow one to analyze, for example, a wide variety of range capabilities for weapon systems in Bonder's [4,6] constant-speed attack model (5). These tables have been calculated by the recursive methods given in Section 8 of Taylor and Brown [24].

A representative tabulation of the hyperbolic-like LCS functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and $T_{\alpha}(x)$ for $\alpha=3/5$, similar to those that appear in references 25 and 26, is given in Tables V and VI. The values of the argument x are the same as those used for the tabulation of the hyperbolic functions by ABRAMOWITZ and STEGUN^[1]. We observe from Table VI and (52) that the limiting value of $T_{\alpha}(x)$ as $x \to +\infty$ (here $\alpha=3/5$) is quickly reached, with three-decimal-place agreement by x=4.5.

7. NUMERICAL EXAMPLES

In this section we examine a couple of numerical examples to show some of the insights that may be gained into the dynamics of combat between two homogeneous forces from our new results. As in references 21 and 24, we consider S. Bonder's [4,6] model (5) for the constant-speed attack against a static defensive position. We will focus on the new results of this paper [in particular, the prediction of battle outcome from initial conditions without explicitly computing the force-level trajectories (cf. questions (Q1) and (Q4) of Section 1)]. From the input data given in Table VII, we

T3/5(x)	1.17630 1.18172 1.19731 1.19731	1.20258 1.20258 1.21253 1.221738	1.22687 1.23150 1.24055 1.24496	1.25930 1.25778 1.25778 1.26191	1.26998	1.298264 1.29620 1.29620 1.30914	1.3000983 1.31000983 1.316346 1.316946	1.32261 1.32566 1.32866 1.33161 1.33451	1.33 736 1.34 016 1.34 562 1.34 562	1.355089 1.3558459 1.3558459 1.3558459	1.36327
H _{2/5} (x)	1.70597 1.72536 1.76489 1.76486	1.000431 1.000431 1.006463 1.006500	1.90621 1.92702 1.94802 1.96917 1.99047	22-00-00-00-00-00-00-00-00-00-00-00-00-0	2-12175 2-14423 2-16688 2-18972 2-1274	2-23994 2-23994 2-282994 2-382994 2-382994 2-382994	20000000000000000000000000000000000000	2.50394 2.50394 2.52956 2.55956 2.55956 4.00	2.65415 2.65082 2.66082 2.68175 2.1490	2-14229 2-16922 2-19119 2-82589 2-82589	2.86285
F3/5(x)	1	1.50037 1.52131 1.53198 1.54278	1.5554 1.5554 1.5554 1.5554 1.5554 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5555 1.5	1.61044 1.62221 1.63411 1.64616 1.65836	1.66318 1.66318 1.69581 1.70859 1.72152	1.73460 1.74784 1.7477 1.78847	1.000233 1.000233 1.000524 1.00052 1.00052	1.834602 1.90384 1.91901 1.91901	1.96.98 1.96.53 1.99138 2.0138	2.03000 2.04656 2.06331 2.09735	2,11465
×	00000 00000 000004	00000	0-24 0-24 0-24 0-24	20×00	0-264 0-264	1.226			111111 1 0 0 0 0 0 1 0 1 0 0 0 0 1 0 1 0	44444 44444 8040-00	1.50
				•							
T _{3/5} (x)	0.77922 0.78994 0.80054 0.81100 0.82134	0.84166 0.85165 0.85163 0.86148	0.88081 0.89030 0.8968 0.90893 0.91807	0.93599 0.94478 0.95478 0.95346	0.97048 0.98705 0.99517 1.00318	1.01109 1.02659 1.03416 1.0416	1.056903 1.05690 1.06948 1.07055	1.008440 1.009118 1.10445 1.10445	1.012363 1.012363 1.012586 1.013596	1.15477 1.15953 1.16521 1.10680	1.17630
H _{2/5} (x)	0.86199 0.887731 0.90803 0.92855 0.92343	0.998887 0.96987 0.96987 1.00000	1.01670 1.03240 1.06815 1.06896 1.07983	1.09575 1.11174 1.12778 1.14389	1.17632 1.209034 1.22550 1.22550	1.25866 1.297534 1.30903 1.30903	1.344004 1.344004 1.34462	1.42971 1.44734 1.46508 1.48292 1.50086	1.51892 1.55537 1.55537 1.57377	1.61092 1.62968 1.64856 1.66757	1.70597
F _{3/5} (x)	1.110622 1.11060 1.11507 1.11963	1.12905 1.12905 1.12889005 1.148885	1.15427 1.15960 1.16504 1.17057 1.17620	1.18193 1.18776 1.19370 1.20587	1.21211 1.224895 1.234895 1.23145	1.254486 1.255173 1.255870 1.26579	1.288028 1.29521 1.30284 1.310584	1.3200 1.3200 1.32466 1.35066 0.066	1.35942 1.357942 1.37663 1.38541	10.04033 10.04033 10.0421748 10.0431748 10.0431748	1.45028
×	00000 0~0000 0~000	00000 WWWW W 0 ~ 20	00000 04000 04004	00000	00000	00000	00000 888888 0-12844	00000 88888 89898 89898	00000 00000 00000	00000	1.00
T _{3/5} (x)	0.0 0.03607 0.06279 0.08684	0.13063 0.15113 0.17088 0.19008	0.22703 0.262410 0.26241 0.27959	0.31300 0.345944 0.345954 0.36141 0.37706	0.39250 0.40773 0.42276 0.43761	0.46675 0.49520 0.50917 0.52298	0.559662 0.55915 0.563415 0.57664 0.57664	0.60251 0.62791 0.64037 0.65268	0.65486 0.67690 0.68890 0.70057	0.72370 0.73506 0.75740 0.75740	0.77922
H _{2/5} (x)	0.00	0.13076 0.13123 0.17123 0.19058 0.20958	0.22748 0.24613 0.26398 0.28157	0.331603 0.33296 0.34971 0.36631 0.38275	0.39906 0.41525 0.43132 0.44729	0.5478970.520032	0.5545689 0.5543689 0.564369 0.66339	0.64362 0.64387 0.66411 0.67933	0.72494 0.72494 0.75532 0.75532	0.78573 0.80095 0.81618 0.83143	0.86199
F _{3/5} (x)	1.000000 1.0000000 1.000000000000000000	1.000104 1.000504 1.000267 1.00367	1.000417 1.000505 1.000601 1.000601	1.00939 1.010659 1.01353 1.01353	1.01672 1.01844 1.02024 1.02213	1.02617 1.02632 1.03055 1.03287	1.03776 1.04034 1.04301 1.04576 1.04860	1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055 1.055	1.06750 1.07697 1.07452 1.0816	1.08572 1.08963 1.09364 1.09774	1.10622
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Lanchester-Clifford-Schläfli Functions $F_{\alpha}(x)$, $H_{1-\alpha}(x)$, and from 0.00 to 1.50. $T_{\alpha}(x)$ for Table V.

and

 $\alpha = 3/5$

a = 3/5	T _{3/5} (x)	1. 488949 1. 488949 1. 488949 1. 488949	111111 14444 14444 168888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 16888 1688	11111 111111 1111111111111111111111111	44444 98888 99999 90000 00000	11111 4444 900000 9000000000000000000000	11111 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444 4444	11111 4448 4484 4484 4484 4484 4484 448	1. 48951 1. 48951 1. 48951 1. 48951 1. 48951	1.48951	
	H _{2/5} (x)	228.05212 251.58239 277.54946 306.20625 337.83198	372.73492 411.25538 453.76907 500.69077 552.47842	609.63759 672.72646 742.36131 819.22253	997.70714 1101.07568 1215.17807 1341.13066 1480.16606	1633.64526 1803.07095 1990.10239 2196.57164 2424.50162	2676-12602 2953-91126 3260-58073 3599-14166 3972-91466	4385.56643 4841.14586 5344.112390 5899.43758 6512.53463	7189-44720 7936-81115 8761-97155 9673-03503 10678-95366	11789.61318	
	F _{3/5} (x)	153-13173 168-90493 186-33813 205-57712 226-80937	250.24183 276.10298 304.64504 336.14649 370.91477	409.28931 451.64484 498.39511 549.99691 606.95457	£69.82494 739.22288 815.82701 900.38700 993.73032	1210-51696 1210-51696 1336-68310 1474-69908 1627-72303	1796.65455 1983.14955 2189.03632 2416.33389 2647.27158	2944.31108 3250.17066 3587.85195 3960.66950 4372.28341	4826.73538 5328.48855 5882.47146 6494.12659 7169.46404	7915.12075	
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	T _{3/5} (x)	1.44099 1.45665 1.46549 1.46549	1.47127 1.47722 1.47722 1.48123	1.48272 1.48394 1.48594 1.48577 1.48577	1.4845 1.48745 1.48812 1.48837	11	44444	1+89938 1+89940 1+89940 1+899440	1.44889446 899446 11.44889446 11.44489449	1.48949	
	H _{2/5} (x)	4.69636 5.17096 5.69257 6.26617 6.89726	5918 1967 1764 1764	12.57485 13.51608 14.88417 16.39224 18.05476	19.88770 21.90868 24.13713 26.59449 29.30442	32. 25. 25. 25. 25. 25. 25. 25. 25. 25. 2	52.53584 57.91754 63.45430 70.46357 77.62878	85.59991 94.39423 104.09703 114.80242 126.61432	139.64742 154.02832 169.89679 187.40709 206.72954	228.05212	
	F _{3/5} (x)	3.5559 3.5557 3.5572 4.56725 7.08459 6.3063 6.3063	5.16006 5.66727 6.22710 6.84489 7.52655	8.27860 9.10823 10.02339 11.03285 12.14631	13.37444 14.72906 16.22317 17.87117 19.66890	21.69389 23.90546 26.34495 52.03589	35.27878 38.89107 42.87609 47.27239 52.12255	57.47356 63.37728 65.89096 77.007779 85.00753	93.75716 103.41170 114.026496 125.82052 138.79270	153.10773	
	×	0-0000 0-0000	~~~~~ ~~~~~	MWWWW ••••••	M-@№-@& M-M-M-M-M			พพพพพพ ฉ⊶ผพง		0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	
T _{3/5} (x)	1.366362 1.366362 1.366362 1.370193	3446 3446 3446 3466 3466 3466 3466 3466	1.386496 1.386692 1.386867 1.390074	1.39442 1.39442 1.394621 1.36970 1.40139	1. +00306 1. +00469 1. +00483 1. +00483	11111111111111111111111111111111111111	10.00000000000000000000000000000000000	1. 425567 1. 42590 1. 42831 1. 42831	11 444 444 890 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 11 1	1. 44340 1. 44340 1. 44380 1. 44380 1. 44380 1. 44380 1. 44380	1.44094
H _{2/5} (x)	2.98285 2.94080 2.97015 2.99915	0559 0590 151	8.00 mm m	W	545 580 680 680 680 680 680 680 680 680 680 6	8 - 10 - 10 - 10 - 10 - 10 - 10 - 10 - 1	9704 9704 9463 9847 0234	4.0624 4.16234 4.182534 7.283534	4.30640 4.30640 4.347466 4.387467 4.38767	4444 6502465 6502465 6502465 6502465 6503	4.69636
F _{3/5} (x)	2.1145 2.145 2.14982 2.18740 2.188740	2040 2224 2411 2599	2.29 2.39 2.33 2.33 2.35 2.35 2.35 2.35 2.35	2.441825 2.441825 2.443895 2.453995 2.48113	2.520 2.524 2.524 2.536 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568 2.568	2.61296 2.63574 2.655874 2.655876 7.0553	257 78 80 80	2.85175 2.87701 2.90253 2.95831 2.95436	2.98067 3.00726 3.03412 3.06125 3.06125	3.11.636 3.14636 3.17259 3.20114	3.25912
×	0.00000 0.000000 0.000000	NUNUNU.	00000	29090	7777	17.75	0-266	2/9~86 80 80 80 80 80 80 80 80 80 80 80 80 80	11.99	11111	2.00

Table VI. Lanchester-Clifford-Schläfli Functions $\mathbf{F}_{\alpha}(\mathbf{x})$, $\mathbf{H}_{1-\alpha}(\mathbf{x})$, and from 1.50 to 10.0. × T_{\alpha}(x) for

and

 $\alpha = 3/5$

Table VII. Input Data for Numerical Examples

$$\mu = 1, \nu = 2$$

 $\alpha_0 = 0.06 \text{ X}$ casualties/minute/Y unit

 $\beta_0 = 0.6 \text{ Y casualties/minute/X unit}$

 $R_{\alpha} = R_{\beta} = 2000 \text{ meters}$

v = 5 miles/hour

compute the parameter values shown in Table VIII. We observe from Tables VI and VIII the predicted agreement between $\Gamma(1-\alpha)/\Gamma(\alpha)$ and the limiting value of $T_{\alpha}(x)$ as $x \to +\infty$ (see (52)) for $\alpha = q = 3/5$. We now consider two cases: (I) $R_0 = 2000$ meters, and (II) $R_0 = 1250$ meters.

When $R_0 = 2000$ meters (see Figure 3 of Taylor [21]), we have $K_S = 0$ and τ_0 = 0. The maximum time that the battle can last is t_{max} = 14.91 minutes, since at this time the attackers reach their final objective (i.e. the defensive position). We now consider the qualitative behavior of the $\mu=1$, $\nu=2$ force-level trajectory shown in Figure 3 of Taylor $^{[21]}$. Theorem 2 tells us that X can be annihilated $\approx x_0/y_0 < 0.420$. By (56) the annihilation time of the X force is given by $T_q(\tau(t_X^a)) = 3.544 \times_0/y_0$. For $x_0 = 10$, $y_0 = 30$, we have $T_q(\tau_X^a) = 1.18122$ so that from Table V (using linear interpolation) we obtain τ_{χ}^a = 1.009. Hence, (39) yields t_{X}^{a} = 14.24 minutes and r_{X}^{a} = 89.8 meters. Further results are given in Table IX. When $R_0 = 1250$ meters (see Figure 3 of Taylor and Brown [24]), we have $K_S = 5.5923$ minutes, $\tau_0 = 0.0975$, and $t_{max} = 9.32$ minutes. In this case (again, for $\mu = 1$, $\nu = 2$), X can be annihilated $\Leftrightarrow x_0/y_0 < 0.382$ with [from (54)] the annihilation time of the X force given by $T_q(\tau_X^a) = (3.565 \text{ u}_0 + 0.223)/(0.156 \text{ u}_0 + 1.004)$, where $u_0 = x_0/y_0$. Some further numerical results are given in Table X. Again, these parametric results should be contrasted with the single μ = 1, ν = 2 force-level trajectory shown in Figure 3 of reference 24.

8. Discussion

In Section 7 above we have seen how our new definition of power Lanchester functions (guided by the general requirements for GLF given in Table I) allows one to conveniently obtain much valuable information about the model (1) with attrition-rate coefficients (7) without explicitly computing the entire force-level trajectories (see Note 12). Previously we were limited to only computing force-level trajectories.

Now we can tell who is going to be annihilated and when without explicitly computing the trajectories (see Note 13). Not only did we answer questions about qualitative

Table VIII. Parameter Values for Numerical Examples

$$k_a = 4.0233 \times 10^{-3} \text{ X casualties/(minute)}^{\mu}/\text{Y} \text{ unit}$$

$$k_b = 2.6979 \times 10^{-3} \text{ Y casualties/(minute)}^{V}/\text{X unit}$$

$$p = 2/5, q = 3/5$$

$$\Gamma(p)/\Gamma(q) = 1.48951$$

$$K_0 = 0$$

Table IX. Annihilation of the X Force as a Function of the Initial Force Ratio for $R_0 = 2000$ meters

(x_0/y_0)	<u>t^a(minutes)</u>	r_{X}^{a} (meters)
0.333	14.24	89.8
0.250	11.61	443.2
0.200	10.19	633.2

Table X. Annihilation of the X Force as a Function of the Initial Force Ratio for $R_0 = 1250$ meters

(x_0/y_0)	<u>t</u> X(minutes)	$\frac{r_X^a(meters)}{}$
0.333	10.63	†
0.250	7.56	235.9
0.200	6.17	422.8

 $t_{\text{max}} = 9.32 \text{ minutes} \text{ and } x_{\text{f}} = x(r=0) = 1.35.$

model behavior (e.g. force annihilation) for specific values of, for example, initial force levels but also for a range of values of the initial force ratio (i.e. parametric analysis of model behavior). The results of this paper may be used for other parametric analyses (see Note 14), e.g. parametric dependence of battle outcome on attrition-rate coefficients. Thus, our extension of past results [24] allows one to develop important insights into the dynamics of combat between two homogeneous forces with temporal variations in fire effectivenesses. With the availability [25,26] of tabulations of the LCS functions, one can now analyze such combat modelled by the power attrition-rate coefficients (7) with somewhat the same facility as he can for the constant-coefficient case (see Note 15) and thus aid in parametric analyses.

In his classic 1914 paper [17] Lanchester assumed that the combatants' fire effectivenesses (as expressed by Lanchester attrition-rate coefficients) were constant over time and deduced his famous square law

$$\beta\{x_0^2 - x^2(t)\} = \alpha\{y_0^2 - y^2(t)\}, \tag{57}$$

where α and β denote constant attrition-rate coefficients. It follows from (57) that (provided there is no "time limit" for the battle)

X will be annihilated
$$\approx x_0/y_0 < \sqrt{\alpha/\beta}$$
. (58)

Thus, we see that equality of Lanchester-type fighting strengths depends on two parameters: (I) initial force ratio, and (II) relative effectiveness. When the timing of military actions is considered, we add a third parameter, the intensity of combat = $\sqrt{\alpha\beta}$, to this list of significant combat parameters. No such simple relationship like the square law (57), which yields (58), holds in general for variable attrition-rate coefficients when $a(t)/b(t) \neq constant$. However, by transforming the independent variable t to normalize the battle's time scale by the intensity of combat, we found (see equation (37)) that the course of such variable-coefficient combat depends on only two weapon-system parameters: (I) relative fire effectiveness, R(t) = b(t)/a(t), and (II) intensity of combat, $I(t) = \sqrt{a(t)b(t)}$. Moreover, we extended (58) to combat

modelled with the power attrition-rate coefficients with "no offset" (7) (see Theorem 2). This is the <u>first</u> time that such a generalization of the square law has been obtained for the variable-coefficient Lanchester-type model (1) with $a(t)/b(t) \neq constant$. We observe that for $K_S > 0$ this "exact" outcome-prediction relation (i.e. necessary and sufficient condition for force annihilation) involves higher transcendental functions (here, the LCS functions) and is complementary to the sufficient condition (involving only elementary functions) given by Taylor and Parry [28] for $K_S > 0$.

Work by BONDER^[5,7], Clark^[12], and others^[2,9] on the prediction of Lanchester attrition-rate coefficients (see Taylor and Brown^[24] for further discussion and references) has generated interest in variable-coefficient Lanchester-type models. Interest in the power attrition-rate coefficients with "no offset" (7) is provided by S. Bonder's^[4,6] model (5) and his examination of predicted attrition-rate for various weapon systems (see pp. 196-200 of reference 9). However useful our results may be in their own right, they have far greater import: (I) they are a model for the treatment of other Lanchester functions and their tabulations, and (II) they may be used in the numerical determination of the parity-condition parameter^[27] Q* for related attrition-rate coefficients (for example, (4) with $K_0 > 0$). In the future we will show how our tabulations of the LCS functions play a key role in the numerical determination of the parity-condition parameter Q* for the general power attrition-rate coefficients (4) with positive "offset" (i.e. $K_0 > 0$).

We have extended our mathematical theory [24] of variable-coefficient Lanchester-type equations of "modern warfare" for combat between two homogeneous forces in order to be able to more thoroughly analyze such models. The classic ordinary differential equation theories (see, for example, HILLE [15]) were inadequate to supply all the answers sought about such combat models (cf. questions (Q1)-(Q4) in Section 1 above). The mathematical theory of the model (1) with coefficients (7) is now nearly as complete as that of the constant-coefficient model. Such results as we have given here are very useful for understanding the dynamics of combat, i.e. how the trading of casualties

will be projected over time. H. K. WEISS^[30] has emphasized that such a simplified model of a combat situation is particularly valuable when it leads to a clearer understanding of significant relationships that would tend to be obscured in a more complex model. As is always the case, however, the insights geined into combat dynamics are no more valid than the models themselves.

9. SUMMARY

In this paper we have introduced new mathematical functions (Lanchester-Clifford-Schläfli, or LCS, functions) that allow important information (in particular, forceannihilation prediction) to be obtained without explicitly computing force-level trajectories for the variable-coefficient Lanchester-type model (1) with power attrition-rate coefficients with "no offset" (7). Our development was based on new theoretical considerations: we gave a new general discussion of representing the solutions to the X and Y force-level equations in terms of general Lanchester functions (GLF) and gave the general properties that these GLF should possess; we showed that there are essentially only two kinds of GLF that satisfy these requirements (exponential-like GLF and hyperbolic-like GLF) and that the hyperbolic-like functions are to be preferred. Moreover, the exponential-like GLF are an essential theoretical construct, since they play a key role in determining force-annihilation-prediction conditions (i.e. showing that the reciprocal of the parity condition parameter is equal to the limiting value of the quotient of two hyperbolic-like general Lanchester X-functions). We stressed that such building blocks should be chosen to yield as much information as possible about the model (and as conveniently as possible). We saw that the analysis of, for example, the X force-level equation was facilitated by transforming the battle's time scale and that the only two weapon-system parameters affecting the course of combat are the relative fire effectiveness and the intensity of combat. These results extended and unified our mathematical theory of variable-coefficient Lanchester-type equations of "modern warfare" (see reference 24).

We then applied our general mathematical theory to the special case of combat modelled by power attrition-rate coefficients with "no offset." Our new definition of Lanchester-Clifford-Schläfli (LCS) functions was required for these power attrition-rate coefficients in order to answer questions about battle outcome without explicitly computing force-level trajectories (i.e. to predict battle outcome/force annihilation). The mathematical theory of this variable-coefficient Lanchester-type models of "modern warfare" (modelling, for example, weapon systems with the same effective range) is now nearly as complete as that of the constant-coefficient model. With tabulations of the new LCS functions now available, one can study this variable-coefficient model almost as easily and thoroughly as Lanchester's classic constant-coefficient model.

NOTES

- 1. Following terminology introduced in reference 24, we will refer to Lanchester functions corresponding to the power attrition-rate coefficients (4) with $K_0 > 0$ as offset power Lanchester functions (see Section 1). The power Lanchester (i.e. LCS) functions correspond to $K_0 = 0$, i.e. to the power attrition-rate coefficients with "no offset" (7).
- 2. The equations (1) are only valid for x,y > 0. The first, for example, becomes dx/dt = 0 for x = 0.
- 3. Further information on sets of circumstances that have been hypothesized to yield the combat equations (1) (with constant coefficients) may be found in BRACKNEY^[11] and Weiss^[29].
- 4. It is impossible for all a(t), b(t) > 0 satisfying Condition (A) to have $x_i, y_i > 0$ for i = 1, 2 and all $t > t_0$ such that $\dot{x}_i = -\sqrt{k_b/k_a} \ a(t)y_i$ and $\dot{y}_i = -\sqrt{k_a/k_b} \ b(t)x_i$ and x_1, x_2 are linearly independent. If it were possible, then $|S(t)| = -x_1y_2 + x_2y_1 = C_S$ and without loss of generality we may take $C_S > 0$.

Introducing $u_i = x_i/y_i$ and $\Delta = u_2 - u_1$, we would have $|S(t)| = y_1y_2\Delta = C_S > 0$ and $\Delta = \sqrt{k_b/k_a} b(t)(u_1+u_2)\Delta > \sqrt{k_b/k_a} b(t)\Delta^2 > 0$ for $t > t_0$ so that $\Delta(t) > 0$ is strictly increasing for $t \ge t_0$. It would follow that $1/\Delta(t_0) - 1/\Delta(t) \ge \sqrt{k_b/k_a} \int_{t_0}^t b(s)ds$, which is impossible for b(t) such that $\lim_{t\to +\infty} \int_{t_0}^t b(s)ds = +\infty$.

- 5. To keep x_1 , x_2 , y_1 , and y_2 as "simple" as possible we specify that their initial values at t_0 to be either 0 or 1. In order that $|S(t)| = x_1 y_2 x_2 y_1 \neq 0$, we must therefore have either $x_1(t=t_0) = y_2(t=t_0) = 0$ and $x_2(t=t_0) = y_1(t=t_0) = 1$, or $x_1(t-t_0) = y_2(t=t_0) = 1$ and $x_2(t=t_0) = y_1(t=t_0) = 0$. We consider the first possibility with similar arguments holding for the second. In this first case examination of the differential equations with initial conditions shows us that $x_1 = S_X$, $x_2 = C_X$, $y_1 = C_Y$, and $y_2 = S_Y$, i.e. the functions coincide with the hyperbolic-like GLF of Table III. Thus, we need not consider L = I, since the same results may be obtained by using another one of Lemma 1's feasible forms for L.
- 6. We conjecture that some condition like $\lim_{\tau \to +\infty} \tau(t) = +\infty$ is sufficient to guarantee that Q* is unique.
- 7. This intuitively obvious result may be proved by observing the identity

$$\int_{s}^{t} \{b(\sigma)x^{2}(\sigma) + a(\sigma)y^{2}(\sigma)\}d\sigma = x(s)y(s) - x(t)y(t).$$

A less obvious fact is that unless at least one of $\int_0^\infty a(t)dt$ and $\int_0^\infty b(t)dt$ is unbounded, then neither x(t) nor y(t) need ever be annihilated (see Hille [15]).

As an example of this situation, consider the battle with attrition-rate coefficients (7),

 $K_S > 0$, and $\mu = \nu < -1$. Then $x(t) = x_0 \cosh \theta(t) - y_0 \sqrt{k_a/k_b} \sinh \theta(t)$, where $\theta(t) = \{-1/(\nu+1)\}\{1/K_S^{-(\nu+1)} - 1/(t+K_S)^{-(\nu+1)}\}$. Let $\theta_\infty = \{-1/(\nu+1)\}\{1/K_S^{-(\nu+1)}\} > 0$ and finite. It follows that $\nu < -1$ and $K_S > 0$ can be chosen so that even though

 $x_0 < y_0 \sqrt{k_a/k_b}$ we have $\lim_{t \to +\infty} x(t) = x_0 \cosh \theta_{\infty} - y_0 \sqrt{k_a/k_b} \sinh \theta_{\infty} > 0$.

- 8. In general, the value of Q* will not be known exactly. Unfortunately, errors in the initial conditions for the exponential-like GLF become exponentially magnified over time. The situation is even worse for $\eta_E(t;Q*) = E_{2X}^+(t;Q*) = E_{X}^+(t;Q*)/E_{X}^-(t;Q*)$, which is used to determine the time that the X force will be annihilated.
- 9. The term <u>quasi-autonomous</u> was coined by Taylor^[22] (<u>see</u> also TAYLOR^[23]) to denote a system of differential equations that may be transformed to an autonomous system (<u>see</u> for example, p. 163 of PETROVSKI^[18]) by a change of the time scale. Special cases of such Lanchester-type equations have been considered by, for example, Farrell^[9] and TAYLOR^[20]. More general (possibly nonlinear) quasi-autonomous Lanchester-type equations have been studied by Taylor^[22], [23] (<u>see</u> also Note 4 of Taylor and Brown^[24]).
- 10. If we were to specify the initial conditions of the GLF at t=0 instead of $t=t_0$, then (32) would reduce to $x(t)=x_0\tilde{C}_X(t)-y_0\sqrt{k_a/k_b}$ $\tilde{S}_X(t)$. However, when the initial conditions for the hyperbolic-like GLF are not given at t_0 , a separate tabulation of, for example, $\tilde{C}_X(t)$ must be used for each different value of t_0 (i.e. $\tilde{C}_X=\tilde{C}_X(t;t_0)$).
- 11. Although the solution to the X force-level equation (2) with the power attrition-rate coefficients (7) may be expressed in terms of known higher transcendental functions $(\underline{see} \text{ Taylor}^{[21]})$, Taylor and Brown [24], and Taylor and Comstock [27]), we have chosen to introduce the LCS functions, since tabulations of these other functions are not available for the full range of parameter values of interest in Lanchester combat theory. For example, we can construct such solutions with modified Bessel functions of the first kind of fractional order, but tabulations of these (\underline{see} , for example, Abramowitz and Stegun [1]) only exist for a restrictive set of values of the order p (i.e. $p = \pm 1/4$, $\pm 1/3$, $\pm 1/2$, $\pm 2/3$, $\pm 3/4$), where $p = (\mu+1)/(\mu+\nu+2)$. Furthermore, tabulations of functions corresponding to the quotient of, for example, two GLXF do not apparently exist at all. Consequently, we have introduced our new LCS functions, which provide much of the information desired about such battles. The naming of our LCS functions follows from the facts that a function similar to $F_{\alpha}(x)$ was introduced

- by LUDWIG SCHLÄFLI^[19] (1814-1895) in 1867, while a related one appears in a posthumous fragment of the great English geometer William Kingdon Clifford (1845-1879) (see pp. 343-348 of CLIFFORD^[13]).
- 12. In his well-known survey paper on the Lanchester theory of combat, Dolansky [14] suggested the development of outcome predicting relations without solving in detail and/or computing force-level trajectories as one of several problems for future research. Our Theorem 2 is a step towards this problem's resolution (see also references 22, 27, and 28).
- 13. Bonder and Honig^[10] point out, however, that force annihilation may not be the appropriate criterion for evaluating many military operations, especially when force annihilation does not occur. See pp. 192-242 of Bonder and Farrell^[9] for a detailed Lanchester-type analysis of an attack situation for which other "end of battle conditions" play the major role in the evaluation process. Nevertheless, it is of interest to know when and why force annihilation will occur.
- 14. S. BONDER^[8] has suggested that an increased emphasis be placed on parametric analyses in systems analysis studies (<u>see</u> pp. 21-22 of reference 8).
- 15. One significant exception is that the outcome of fixed-force-level-breakpoint battles (for example, Y "wins" when $x_f = x(t_f) = x_{BP}$ but $y_f > y_{BP}$, where t_f , x_f , y_f denote final values and x_{BP} denotes X's breakpoint) with x_{BP} , $y_{BP} > 0$ and $a(t)/b(t) \neq constant cannot apparently be analyzed in the manner described in this paper (see Taylor and Comstock [27]).$

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